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## Origins Spectral Interpretation Resource Identification Security-Regolith Explorer (OSIRIS-REx) Project

# OSIRIS-REx Touch-and-Go Camera Suite (TAGCAMS) Data Product Software Interface Specification

UA-SIS-9.4.4-322, Rev. 5.0

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#### CM FOREWORD

This document is an OSIRIS-REx Project controlled document. Changes to this document require prior approval of the OSIRIS-REx Configuration Control Board (CCB) and Configuration Management Lead (CML). Proposed changes shall be submitted to the OSIRIS-REx Project CML, along with supportive material justifying the proposed change.

Questions or comments concerning this document should be addressed to:

SPOC Configuration Management Team  $1415\ N.\ 6^{th}\ Avenue$   $Tucson,\ AZ\ 85705$ 

Email: <a href="mailto:spoc-cm@orex.lpl.arizona.edu">spoc-cm@orex.lpl.arizona.edu</a>

# OSIRIS-REx Project TAGCAMS Data Product SIS

SIGNATURE PAGE

| Prepared By:                                      |                |
|---|----------------|
| M. Katherine Crombie OSIRIS-REx PDS Lead          | Date           |
| Approved By:                                      |                |
| Heather Enos<br>OSIRIS-REx Deputy Principal Inve  | Date estigator |
| Brent Bos TAGCAMS Instrument Scientist            | Date           |
| Michael Moreau<br>OSIRIS-REx Flight Dynamics Lead | Date<br>d      |
| Coralie Jackman OSIRIS-REx Flight Dynamics        | Date           |
| Karl Harshman<br>OSIRIS-REx SPOC Manager          | Date           |
| Sanford Selznick OSIRIS-REX Science Processing L  | Date           |

## DOCUMENT CHANGE LOG

| REV/VERSION | DESCRIPTION OF CHANGE  | APPROVED | DATE       |
|-------------|--|----------|------------|
| LEVEL       |  | BY       | APPROVED   |
| 1.1         | Initial Release  |          | 06/08/2015 |
|             | Update Status Packet values from Last Opcode to DVR +5V from 8-bit to 32- bit to correct error in Version 1.0 SIS Clarify description of L1 Status data product Add Status L1 data product specification Make data format descriptions consistent with TAGCAMS Users Guide 4/9/2015 In Image Format, break ATT_QUAT keyword into 4 keywords, break ATT_RATE keyword into 3 keywords Table 2. Insert row for Instrument Alignment and Calibrations Updated signature page Updated data formats descriptions consistent with TAGCAMS Users Guide July1, 2105 Added MID-OBS, DELTAOBS, INST_QA, INST_QX, INST_QY, INST_QZ, RADESYS, EQUINOX, CKQUAL, MISSPACK, CHCKSUM keywords to image header Removed TARGET, MPHASE, ACTIVITY, ATLTGTID, SCISEQID, DESCRIPT, OBJECT, OBJECTRA, OBJECTDEC, COORDSYS, LAT, and LONG keywords from the image FITS header information. Changed spacecraft quaternion keywords in the image FITS header from SC_Q0, SC_Q1, SC_Q2, SC_Q3 to SC_QA, SC_QX, SC_QY, SC_QZ Change ET time to be mid-observation time instead of start-observation time in the image FITS header Update section 4.3.2 to update OPNAV image delivery method to the FOB Removed reference to PDS Label |          |            |
| I           | Example  |          |            |

| 2.1 | Update INSTRU keyword to INSTRUME (FITS Standard) Update per ECR-0047                                     | 08/10/2016 |
|-----|---|------------|
| 2.2 | Update per ECR-0061 Updated D-PI from Ed Beshore to Heather Enos Removed Ground Segment Manager signature | 11/15/2016 |
| 3.0 | Update per ECRs and changes from 2.n update cycle   | 03/09/2017 |
| 4.0 | Update per CR-144   | 05/03/2018 |
| 5.0 | Update per CR-323   | 10/23/2018 |

## LIST OF TBDs/TDRs

| SECTION ID | DESCRIPTION OF TBD/TBR  | DATE OF    |
|------------|---|------------|
|            |   | RESOLUTION |
| 2          | OSIRIS-REx Archive Volume SIS (Deprecated document, removed from applicable documents)                  |            |
| 4.4.3      | Coordinate System Document Reference  | 9/1/2015   |
| 7.5        | PDS Label Examples (to be completed after Version 1.2 is signed) Removed this document section 9/1/2015 | 9/1/2015   |
| 4.3.3      | Table 4. Data Volume – to be updated with latest rev of the DRM   |            |
| 4.2        | Rate of status packet data collections  | 6/5/2015   |
|            |   |            |
|            |   |            |
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## 1 Purpose and Scope

The data products described by this Software Interface Specification (SIS) are the OSIRIS-REX Touch-and-Go Camera Suite (TAGCAMS) raw and uncalibrated data products. TAGCAMS consists of three similar camera heads: NavCam 1, NavCam 2 and StowCam, each suited for a particular purpose. NavCam 1 is the primary Navigation Camera (NavCam) and is a wide-angle framing camera used for optical navigation. NavCam 2 is the primary Natural Feature Tracking Camera (NFTCam) used for landmark identification and autonomous feature tracking to aid in sample acquisition. Finally, the Stowage Camera (StowCam) is used to document the insertion of the sample collection head into the Sample Return Capsule (SRC). The OSIRIS-REx Science Processing and Operation Center located at the University of Arizona produces these data products and distributes them to the OSIRIS-REx Flight Dynamics Team, the Lockheed Martin Mission Support Area, the Science Team and the Planetary Data System. The purpose of this document is to provide users of these data products with detailed descriptions of the products and how they were generated, including data sources and destinations. The document is intended to provide enough information to enable users to read and understand the data products. The users for whom this document is intended are the flight dynamicists and mission operations personnel who will use the data, and the scientists who will analyze the data, including those associated with the project and those in the general planetary science community.

## 2 Applicable Documents and Constraints

This Data Product SIS is consistent with the following Planetary Data System Documents:

- 1. Planetary Data System Standards Reference, Version 1.7.0, September 15, 2016.
- 2. PDS4 Data Dictionary Abridged Version 1.7.0.0, September 28, 2016.
- 3. PDS4 Information Model Specification, V.1.7.0.0, September 28, 2016.

This Data Product SIS is responsive to the following OSIRIS-REx documents:

- 4. OSIRIS-REx Science Data Management Plan, UA-PLN-9.4.4-004, Rev 4.0, May 26, 2016.
- OSIRIS-REx Science Processing and Operations Center and Planetary Data System Small Bodies Node Interface Control Document, UA-ICD-9.4.4-101, Rev 1.0, October 2013.
- 6. OSIRIS-REx Mission Support Area and Science Processing and Operations Center Interface Control Document, NFP3-PN-12-OPS-6A.
- 7. OIA ORX 092, OPNAV Images (NavCam and OCAMS), April 7, 2015.
- 8. OSIRIS-REX TAGCAMS Users Guide, MSSS-TAG-REQ-4401, August 18, 2016.

- 9. OSIRIS-REx Science Processing and Operations Center and Flight Dynamics Interface Control Document UA-ICD-9.0.0-100, Rev 3.0, June 1, 2016.
- 10. OSIRIS-REx Coordinate System for Bennu, Version 2.0, January 14, 2016.

Finally, this SIS is meant to be consistent with the contract negotiated between the OSIRIS-REX Project and the Science Processing and Operations Center.

## 3 Relationship with Other Interfaces

Changes to the data products described in this SIS effect the following software, products or documents:

**Table 1 - Interface Relationships** 

| Name                    | Type     | Owner   |  |
|-------------------------|----------|---------|--|
| SPOC Database Schema    | Product  | SPOC    |  |
| NavCam Image Data       | Product  | SPOC    |  |
| NFTCam Image Data       | Product  | SPOC    |  |
| STOWCam Image Data      | Product  | SPOC    |  |
| TAGCAMS Raw Status      | Product  | SPOC    |  |
| Data                    |          |         |  |
| TAGCAMS Processed       | Product  | SPOC    |  |
| Status Data             |          |         |  |
| SPOC Archive Packager   | Software | SPOC    |  |
| MSA-SPOC ICD            | Document | MSA     |  |
| SPOC-FDS ICD            | Document | SPOC    |  |
| OSIRIS-REx Science Data | Document | Project |  |
| Management Plan         |          |         |  |

## 4 Data Product Characteristics and Environment

#### 4.1 Instrument Overview

The OSIRIS-REx Touch-and-Go Camera Suite (TAGCAMS) is a framing imaging system to be used for navigation and engineering support imaging on the OSIRIS-REx asteroid sample return spacecraft. The instrument is provided by Malin Space Science Systems (MSSS) and is a configuration of the MSSS commercial ECAM system with custom software and optics. The instrument consists of two redundant DVR4s, each with a single M50 navigation camera head, NavCam and NFTCam respectively. The DVR with the NFTCam also has a C50 camera head (StowCam) for viewing sample stowage. Supplemental information about the generic specifications of the camera system can be found at <a href="http://www.msss.com/space-cameras/">http://www.msss.com/space-cameras/</a>.

## 4.1.1 Observational Profile and Data Acquisition

Each instrument aboard OSIRIS-REx has specific scientific/engineering objectives in support of the overall mission objective of collecting and returning to Earth a pristine sample of the asteroid Bennu. Instrument specific observation campaigns have been outlined for each phase of the mission to support sample site selection and overall Bennu characterization. The TAGCAMS observation profile is as follows:

| Table 2 | - NavCam | Observation | Profile |
|---------|----------|-------------|---------|
|         |          |             |         |

| Mission Phase      | Observation Campaign Description                |
|--------------------|---|
| Outbound Cruise    |   |
|                    | Instrument Health Check (All)                   |
|                    | Operational Performance (All)                   |
|                    | Instrument Alignment and Calibrations (All)     |
|                    | Earth-Moon Flyby Observations (NavCam)          |
| Approach           |   |
|                    | Optical Navigation Imaging (NavCam)             |
| Preliminary Survey |   |
|                    | Optical Navigation Imaging (NavCam)             |
| Orbit A (1.5km)    |   |
|                    | Optical Navigation Imaging (NavCam)             |
|                    | Instrument Alignments and Calibrations (All)    |
| Detailed Survey    |   |
|                    | Optical Navigation Imaging (NavCam)             |
| Orbital B (1.0km)  |   |
|                    | Optical Navigation Imaging (NavCam)             |
|                    | Radio Science Gravity Field Monitoring (NavCam) |
|                    | Instrument Alignments and Calibrations (All)    |
| Reconnaissance     |   |
|                    | Optical Navigation Imaging (NavCam)             |
| TAG-Rehearsal      |   |
|                    | Optical Navigation Imaging (NavCam)             |
|                    | Natural Feature Tracking (NFTCam)               |
| Sample Collection  |   |
|                    | Optical Navigation Imaging (NavCam)             |
|                    | Natural Feature Tracking (NFTCam)               |
|                    | Sample Stow imaging (STOWCam)                   |

#### 4.2 Data Product Overview

This SIS describes image and instrument status (engineering) data acquired by TAGCAMS. Primary Optical Navigation and Natural Feature Tracking Images are stored as binary Flexible Image Transport System (FITS) files. STOWCam and other images are stored as JPEGs. Engineering Status data are stored as a single binary table file per day. Status records are acquired every 120 (default) seconds. The default value is expected to be updated in-flight to support more frequent aliveness checking. A value of 5 seconds may be more typical. The data products described by this SIS are:

- 1. TAGCAMS Raw Images These images are reconstructed science packet telemetry with immediately received associated timing and spatial information in a FITS format. These images are found in the data\_raw collection.
- 2. TAGCAMS JPEG Images These images are the natively downlinked data primarily from the StowCam that are used to record the stowage of the

sampler head in the sample return capsule. It is possible that NavCam or NFTCam images may be downlinked as JPEGs, however these images would not be used for Optical Navigation or Natural Feature Tracking purposes and would be archived at the conclusion of the mission as supplemental information. The comment section of SPOC generated JPEG images contains the same information that is normally found in FITS image headers. Should we have any of these images to archive, they would be found in the data\_supplemental collection.

- 3. TAGCAMS Raw Status Raw DN value of 48 channels of camera status information. These products are found in the data hkl0 collection.
- 4. TAGCAMS Processed Status –Processed (DN to engineering unit) status information in physical units. These products are found in the data\_hkl1 collection.

## 4.3 Data Processing

All OSIRIS-REx mission science data processing is performed at the University of Arizona Science Processing and Operations Center (SPOC). In addition to science processing, the SPOC stores and processes spacecraft engineering camera suite (TAGCAMS) images to standard outputs for further processing by the engineering and science team.

TAGCAMS image and status telemetry are received by the SPOC via the Lockheed Martin Mission Support Area (MSA) and the DSN. TAGCAMS data are reconstructed from telemetry frames (packets) and stored in the SPOC data repository as raw data (OREx Level 0). Level 0 raw status data are then processed to convert digital number values to engineering units resulting in the L1 processed status data product. This product is also stored in the SPOC data repository.

Raw image data are approximately 10.772MB in size. Immediately received spacecraft orientation information (SPICE S/C C-kernels) taken concurrently with the imagery is processed to provide timing and attitude data that is attached to the raw images. Status data are acquired nominally once every 120 seconds (although may be taken more frequently) and are packaged into a single data file per day. Status file size dependent on the number of reading taken per day, but with nominal settings is on the order of 39Kb for both the L0 raw and L1 processed products.

## 4.3.1 Data Processing Level

Table 3 shows the OSIRIS-REx data processing levels of all science data products described by this SIS. Correlation to NASA and CODMAC data processing levels and definitions can be found in Appendix 7.1. Calibration file data processing levels are not discussed, as calibration files require special production techniques.

**Table 3 - TAGCAMS Data Processing Levels** 

| Data Product                                  | NASA<br>Product<br>Level | OSIRIS-<br>REx Data<br>Processing<br>Level | Description   |
|---|--------------------------|--|---|
| TAGCAMS Raw FITS Images                       | Level-0                  | L0   | Reconstructed Telemetry with associated timing and attitude information |
| TAGCAMS JPEG Images                           | Level-0                  | L0   | Reconstructed Telemetry with associated timing and attitude information |
| TAGCAMS Raw Status Binary<br>Table Data       | Level-0                  | L0   | Reconstructed telemetry with engineering DN values                      |
| TAGCAMS Processed Status<br>Binary Table Data | Level-1                  | L1   | DN engineering values converted to physical units                       |

#### 4.3.2 Data Product Generation

As mentioned previously, all OSIRIS-REx science data processing is completed at the SPOC located at the University of Arizona. The decision was made early in the mission lifecycle, that all processing would be centralized to facilitate the relatively quick turnaround needed by the science and operations teams to make tactical decisions about sample site selection. NavCam and NFTCam images will also be processed by the SPOC and made available to FDS and the MSA through the Flight Operations Bucket (FOB) that uses e-mail to notify interested users. FDS and MSA may also use the OSIRIS-REx WebQuery Tool to identify and access NavCam, NFTCam and STOWCam image and status records of interest from the SPOC data repository. Raw NavCam images will be available to FDS within 30 minutes of receipt of the images by the SPOC.

#### 4.3.2.1 LO and L1 Processing

TAGCAMS image and status telemetry are received from the DSN and passed through the LM MSA Front End Data System (FEDS) to the SPOC FEDS. The SPOC ingests, sorts, reconstructs, decompresses (if necessary) and stores telemetry data as raw observational data that includes observations, timing, spatial and spacecraft attitude information. Timing, spatial and spacecraft attitude information are attached to image headers using spacecraft pointing information (quaternions) that has been received from the spacecraft just prior to receipt of image telemetry. This information is in the form of a SPICE C-kernel that is produced at the Navigational and Ancillary Information Facility (NAIF) and made available to the SPOC via the Flight Operations Bucket (FOB). Spacecraft attitude information is also received in the NavCam and NFTCam telemetry via a 72-byte attitude header attached to images. The C-kernel, attitude header and other

timing information is processed by the SPOC to yield the timing, spatial and ancillary information to be attached to the Level 0 raw image headers. A list of all image header values can be found in Section 5.2. Complete image files are sent from the SPOC to the FOB, and interested users are notified by e-mail that images are available

STOWCam images may be received from the spacecraft via telemetry in a JPEG format. These images are received and stored at the SPOC in the same way as the NavCam or NFTCam data are received. The JPEG images will have timing and attitude data attached to the JPEG comment. The included meta-data will be identical to the meta-data provide in the NavCam and NFTCam FITS headers and PDS labels.

TAGCAMS Status data (housekeeping, hk) are processed through the SPOC to sort and record status records as DN values. The status DN values are then converted into physical units (temperatures, voltages, currents) according to instrument specific conversion polynomials (L0 raw to L1 processed processing). The converted values are then stored along with the original values in the SPOC data repository. Once stored in the SPOC data repository L0 and/or L1 status information can be written to files as specified in Section 5.

Once processing has been completed, images are uploaded to the FOB, and FDS and MSA are notified by e-mail that images are ready. Consumers may then also use the OSIRIS-REx WebQuery Tool to identify and download images or status products of interest. The entire process from receipt of image and attitude data from the MSA to e-mail notification takes less than 30 minutes.

#### 4.3.3 Data Flow

Raw and processed data products are built up in sequential data processing steps addressing specific corrections or calibrations. All data products are built from raw telemetry ingested into the SPOC data repository system. The OSIRIS-REx Instrument, Operations, Flight Dynamics and Science Teams access the data repository through a query tool.

Table 4 shows the expected TAGCAMS data collection by camera and mission phase. The number of expected images is specified as well as the expected data volume of the processed data products. (Note: 1944 x 2592 pixels, at 16-bits per pixel plus header information equals 11.04 MB per image raw, raw data volume is currently calculated as this number times the number of images. Also note that the 11.04MB used for volume calculations is slightly larger than current data product size of 10.772MB, and therefore should be thought of as an upper bound)

Table 4 - TAGCAMS Data Products/ Volume by Mission Phase

| Mission<br>Phase   | Launch | Cruise | Approach | Prelim<br>Survey | Orbit A | Detailed<br>Survey | Orbit B | Recon | TAG<br>Rehearsal | Sample<br>Collection |
|--------------------|--------|--------|----------|------------------|---------|--------------------|---------|-------|------------------|----------------------|
| NavCam #<br>Images | 75     | 450    | 100      | 680              | 1054    | 1386               | 191     | 2210  | 924              | 308                  |
| NavCam<br>Raw (MB) | 828    | 4968   | 1104     | 7507             | 11636   | 15301              | 2109    | 24398 | 10201            | 3400                 |

| Mission<br>Phase            | Launch | Cruise | Approach | Prelim<br>Survey | Orbit A | Detailed<br>Survey | Orbit B | Recon | TAG<br>Rehearsal | Sample<br>Collection |
|-----------------------------|--------|--------|----------|------------------|---------|--------------------|---------|-------|------------------|----------------------|
| NFTCam #<br>Images          | n/a    | n/a    | n/a      | n/a              | n/a     | n/a                | n/a     | n/a   | 98               | 203                  |
| NFTCam<br>Raw (MB)          | n/a    | n/a    | n/a      | n/a              | n/a     | n/a                | n/a     | n/a   | 1082             | 2241                 |
| STOWCam #<br>Images         | n/a    | n/a    | n/a      | n/a              | n/a     | n/a                | n/a     | n/a   | n/a              | 270                  |
| STOWCam<br>Raw (MB)         | n/a    | n/a    | n/a      | n/a              | n/a     | n/a                | n/a     | n/a   | n/a              | 2981                 |
| Status # of<br>Observations | 720    | 4320   | 67680    | 14400            | 22320   | 45360              | 43200   | 70560 | 30240            | 16560                |
| Status (MB)                 | 0.14   | 0.86   | 13.54    | 2.88             | 4.46    | 9.07               | 8.64    | 14.11 | 6.05             | 3.31                 |

It is possible that more than one version of the Raw or Processed data products maybe produced. This is not intended to be routine but may occur if one or more calibration files needs to be updated. Any changes to the data processing pipeline are configuration controlled and follow the standard OSIRIS-REx configuration control process. Reprocessed images are identified in the filename (see Section 4.3.4) and are noted as to why re-processing was necessary.

## 4.3.4 Labeling and Identification

OSIRIS-REx science data products are named according to the OSIRIS-REx Naming Conventions Document (UA-HBK-9.4.4-905). The following paragraphs are excerpts of this document that describe how NAVCAM image and housekeeping files are named. The generalized file naming convention is:

The UTC time is the time of data acquisition derived from the spacecraft clock time.

The instrument is one of the following:

**Table 5 - Instrument Abbreviations** 

| Instrument Name | Abbreviation |
|-----------------|--------------|
| NavCam          | ncm          |
| NFTCam          | nft          |
| STOWCam         | sto          |

#### The product type is:

**Table 6 - Data Product Type** 

| Product Type | Definition                         |
|--------------|------------------------------------|
| L0           | Raw Image, reconstructed telemetry |

| LOJ | JPEG Image, reconstructed telemetry     |  |
|-----|---|--|
| LOS | Raw Status Packet                       |  |
| L1S | Status Packet DNs converted to physical |  |
|     | units                                   |  |

The version portion of the file name is a three-digit number indicating the revision number of that particular data product. It should be noted that operational products sent to FDS do not contain the three-digit revision number. This labeling is consistent with operational interface agreements between the SPOC and FDS.

The PDS type file suffix indicates the type of file the data product is. TAGCAMS data products have one of three file type suffixes, .FITS for image files, .DAT for binary status tables or .JPG for JPEGS.

All TAGCAMS image and status files are created with detached PDS labels. The labels are PDS4 compliant XML format labels with the required sections for ARRAYs and TABLE BASE.

Image data products contain headers. The header meta-data are identical for all TAGCAMS image types and contain information about when and how the image was acquired. Data processing status is also indicated in the header.

## 4.4 Standards Used in Generating Data Products

#### 4.4.1 PDS Standards

All data products described in this SIS conform to PDS4 standards as described in the PDS Standards document noted in the Applicable Documents section of this SIS. Prior to public release, all data products will have passed both a data product format PDS peer review and a data product production pipeline PDS peer review to ensure compliance with applicable standards.

All data products are labeled with PDS4 compliant detached XML labels. These labels describe the content and format of the associated data product. Labels and products are associated by file name with the label having the same name as the data product except that the label file has a .xml extension.

Labels are constructed with the PDS4 Product Class, Product\_Observational sub-class. The Product\_Observational sub-class describes a set of information objects produced by an observing system. A hierarchical description of the contents of Product Observational products is:

Product Observational

Identification\_Area - attributes that identify and name an object.

Logical Identifier - name/location of file

Version ID - version of product

Title – Descriptive name of product

Information\_model\_version - version of PDS4 information model used to create product Product\_Class - attribute provides the name of the product class (Product\_Observational) Modification History - attributes describing changes in data product

Observation\_Area - attributes that provide information about the circumstances under which the data were collected.

Time Coordinates - time attributes of data product

Primary\_Results\_Summary - high-level description of the types of products included in the collection or bundle to facilitate data discovery

Investigation\_Area - mission, observing campaign or other coordinated, large-scale data collection attributes

Observing System - observing system (instrument) attributes

Target\_Identification - observation target attributes

Mission\_Area - mission specific attributes needed to describe data product File\_Area\_Observational - describes a file and one or more tagged\_data\_objects contained within the file

File - identifies the file that contains one or more data objects Table Binary - defines a simple binary table.

Information in the preceding paragraphs was distilled from the PDS4 Information Model provided by PDS. Additional information on PDS4 product labels can be found by selecting "How to Approach a PDS4 Data Set" on the Small Bodies Node web site at <a href="http://sbn.pds.nasa.gov">http://sbn.pds.nasa.gov</a>.

#### 4.4.2 Time Standards

Time Standards used by the OSIRIS-REx mission conform to PDS time standards. The spacecraft clock (SCLK) reference is 1/1/2000 12:00:00 UTC, with a minimum range date from 1/1/2010 to 1/1/2030. Onboard time tagging is the standard 32-bit seconds and 16-bit subseconds. The spacecraft clock string reported in various data products contains the spacecraft clock partition at a number before a slash as well as the seconds dot subseconds, e.g. 3/0545586959.34560. It is possible that the seconds portion of the sclk string at the beginning of a science sequence may be noticeably small (seconds <100), this is due to data collections prior to an instrument - spacecraft clock time synchronization. All OSIRIS-REx data products contain both the spacecraft clock time of data acquisition and a conversion to UTC to facilitate comparison of data products. In the case of TAGCAMS spacecraft clock time is given at the start of the acquisition. TAGCAMS data also contain Ephemeris Time (ET) and Ground Receive Time (GRT) to facilitate processing.

#### 4.4.3 Coordinate Systems

All coordinate systems used by the OSIRIS-REx mission conform to IAU standards. A complete discussion of the coordinate systems and how they are deployed in the mission can be found in the document "OSIRIS-REx Coordinate System Plan" (AP-10) archived in the OSIRIS-REx archive mission bundle documents collection. This document is consistent with the coordinate system plans found in other internal project documents:

- 1. PLA-OSIRIS-REx-SC-CDRL-0153, Coordinate Systems Definition Document
- 2. PLA-OSIRIS-REx-SPEC-0010, OSIRIS-REx Trajectory Standards Document

Internal project documents will not be archived but are included here as a reference for project personnel.

## **4.4.4 Data Storage Conventions**

FITS data products are stored according to the FITS 3.0 Standard. Binary data products are stored as big-endian (MSB) binary. Data formats are explicitly described in Section 5 of the document.

#### 4.5 Data Validation

The SPOC has a comprehensive Verification and Validation Plan for all software used at or developed by the SPOC. All software is configuration controlled and any changes made follow the SPOC Configuration Control Plan, which includes substantive testing of changes. During day-to-day production of L0 data products from telemetry, check sums and spot checks are used to validate that software is producing data products correctly.

In addition to software verification and validation, each OSIRIS-REx data product has been peer reviewed for both PDS data format acceptability and scientific usefulness. No changes are expected to data formats after peer review. The SPOC Configuration Control Plan governs any changes, should they be needed.

When data is prepared for submission to the PDS, the SPOC will use automated PDS / mission-provided validation tools for conformance to the PDS4 standards. Validation of the scientific data contained within the NavCam data products will be performed by OSIRIS-REx team members.

## 5 Detailed Data Product Specification

## 5.1 Data Product Structure and Organization

The OSIRIS-REx data archive is organized by instrument. The TAGCAMS portion of the archive is organized with collections for NavCam, NFTCam, StowCam, and Housekeeping (Status). Scientific image data is stored as a 2-part file with a detached PDS label. The detached PDS labels are PDS4 compliant XML label that describes the contents of the image file. See Appendix 7.5 for an example label. The 2-part FITS image file consists of:

- 1. A primary ASCII header of keyword-value pairs
- 2. A primary binary 2-d array (image)

StowCam data may be transmitted from the spacecraft as JPEG images. These images are archived as supplementary information in JPEG format with meta-data attached. A PDS4 compliant XML label that describes the contents of the image file. See Appendix 7.5 for an example label.

L0 and L1 Status Data are stored in the TAGCAMS DATA\_HK\_(L0/1) collections as binary tables with a detached PDD4 compliant XML label. The detached PDS labels describe the specific structure of the binary table. The binary tables contain 53 fields and have fixed-length records of 200 bytes. Status data is packaged as one Earth day's-worth of status records, with a nominal file size of 720 records per day. The number of records

is strictly dependent on the commanded rate of status packet acquisition, with a nominal rate of 1 packet every 120 seconds.

```
The TAGCAMS bundle directory structure is as follows:
orex.tagcams
        data hkl0 – raw level 0 status (housekeeping)
                cruise 1
                ega
                cruise 2
                approach
                preliminary survey
                orbital a.
                detailed survey
                orbital b
                reconnaissance
                rehearsal
                TAG (Touch-and-go)
        data hkl1 – reduced level 1 status (housekeeping)
                cruise 1
                ega
                cruise 2
                approach
                preliminary survey
                orbital a,
                detailed survey
                orbital b
                reconnaissance
                rehearsal
                TAG (Touch-and-go)
        data raw – level 0 raw image products
                cruise 1
                ega
                cruise 2
                approach
                preliminary survey
                orbital a,
                detailed survey
                orbital b
                reconnaissance
                rehearsal
                TAG (Touch-and-go)
        document – TAGCAMS documentation
```

## 5.2 Data Format Descriptions

## **5.2.1 Images**

Optical Navigation and NFTCam image data are stored in FITS file formats with a single header and data unit (HDU). Header keywords are filled as data processing occurs either by the SPOC Ingest/Digest processing or by the SPOC Spatial Generation processing. The FITS image header that contains meta-data describing the conditions under which the image was taken is described in Table 7. The meta-data are also translated into the PDS4 XML label and appears in the Observation Area Class. Descriptions of attributes in the

table below are either abbreviated or truncated in the product FITS files due to line length limitations in the FITS standards. For all image products sample refers to the fastest changing axis, and line refers to the second fastest changing axis.

**Table 7 - Data Format Descriptions: Image Attributes** 

| Attribute Name                   | FITS Keyword | Units | Description   |
|----------------------------------|--------------|-------|---|
| element_array.data_type          | BITPIX       |       | number of bits per data pixel (16 for L0, -32 for L1)   |
| axes                             | NAXIS        |       | number of data axes   |
| axis_array.sequence_number       | NAXIS1       |       | length of data axis 1   |
| axis_array.sequence_number       | NAXIS2       |       | length of data axis 2   |
| n/a - FITS specific              | EXTEND       |       | FITS dataset may contain extensions   |
| n/a - FITS specific              | BZERO        |       | offset data range to that of unsigned short   |
| n/a - FITS specific              | BSCALE       |       | 1= default scaling factor   |
| investigation_area.name          | MISSION      |       | Mission: OSIRIS-REx   |
| observing system.name            | HOSTNAME     |       | Spacecraft hostname Instrument: OSIRIS-REx Navigation Camera (or OCAMS or NFTCAM if either instrument is used for OPNAV purposes) ncm = NavCam, nft = NFTCAM, sto = STOWCAM, map= |
| observing_system_component.name  | INSTRUME     |       | MapCam, pol= PolyCam,<br>sam = SamCam   |
| n/a - FITS specific              | ORIGIN       |       | University of Arizona<br>Science Processing and<br>Operations Center  |
| mission_area.apid                | APID         |       | Spacecraft Application Identification Number used to indicate the type of data packet received from the spacecraft.   |
| mission_area.ground_receipt_time | GRT          |       | Ground Receive Time in coordinated universal time (YYYY-MM-DDThh:mm:ss.sss)  Coordinated Universal Time file was created by SPOC  |
| mission_area.spoc_date           | SPOCDATE     |       | (YYYY-MM-<br>DDThh:mm:ss.sss)   |
| mission_area.creator             | CREATOR      |       | SPOC GIT repository identifier that uniquely identifies code version used to create the data product.   |
| mission area.date of observation | DATE OBS     |       | YYYY-MM-<br>DDThh:mm:ss.sss<br>observation start, Timestamp<br>(in coordinated universal  |

|                          |          | ·   |   |
|--------------------------|----------|-----|---|
|                          |          |     | time) from image acquisition,                   |
|                          |          |     | derived from the second and                     |
|                          |          |     | sub-second values. This is                      |
|                          |          |     | the timestamp at the start of                   |
|                          |          |     | the observation.                                |
|                          |          |     | Spacecraft mid-observation                      |
|                          |          |     | time (YYYY-MM-                                  |
|                          |          |     | DDThh:mm:ss.sss) in                             |
|                          |          |     | coordinated universal time                      |
|                          |          |     | calculated by (DATE_OBS +                       |
| mission_area.mid_obs     | MIDOBS   |     | .5*EXPTIME).                                    |
|                          |          |     | Spacecraft Clock String at                      |
|                          |          |     | start of observation time.                      |
|                          |          |     | SCLK_STR is formatted as                        |
|                          |          |     | clock   |
| mission_area.sclk_string | SCLK_STR |     | partition/seconds.subseconds.                   |
|                          |          |     | Ephemeris Time (seconds                         |
|                          |          |     | past J2000 epoch, TDB -                         |
|                          |          |     | Barycentric Dynamical                           |
|                          |          |     | Time) at the mid-observation                    |
| mission_area.mid_obs_et  | ET       | Sec | time.   |
|                          |          |     | Actual exposure time in                         |
|                          |          |     | seconds, derived from                           |
| mission_area.exposure    | EXPTIME  | Sec | commanded exposure time.                        |
|                          |          |     | Delta between mid-                              |
|                          |          |     | observation time and                            |
|                          |          |     | spacecraft clock string (Mid-                   |
|                          |          |     | observation time)                               |
|                          |          |     | (SCLK_STR timestamp), in                        |
|                          |          |     | seconds used to verify                          |
| mission_area.delta_obs   | DELTAOBS | Sec | exposure time.                                  |
|                          |          |     | Spacecraft quaternion in                        |
|                          |          |     | J2000 (q0 = cos(t/2))                           |
|                          |          |     | obtained from the NAIF                          |
|                          |          |     | provided C kernel. SPICE                        |
|                          |          |     | convention conversion to 3x3                    |
|                          |          |     | matrix transforms vector in                     |
|                          |          |     | spacecraft frame to J2000                       |
|                          |          |     | frame. This value is                            |
|                          |          |     | calculated using the SPICE                      |
|                          |          |     | interface and numbers will be                   |
|                          |          |     | accurate to the accuracy of                     |
|                          |          |     | the SPICE kernels.                              |
|                          |          |     | Calculations will be based on                   |
|                          |          |     | the MIDOBS time. SPICE                          |
|                          |          |     | quaternion standard is that                     |
| geom.qcos                | SC_QA    |     | Q0 is the scalar value.                         |
|                          |          |     | Spacecraft quaternion in                        |
|                          |          |     | $J2000 (q1 = \sin(\frac{1}{2}))$                |
|                          |          |     | obtained from the NAIF                          |
|                          |          |     | provided C kernel. SPICE                        |
|                          |          |     | convention conversion to 3x3                    |
|                          | i        |     | matrix transforms vector in                     |
|                          |          |     |   |
|                          |          |     | spacecraft frame to J2000                       |
|                          |          |     | spacecraft frame to J2000 frame. This value is  |
|                          |          |     |   |
|                          |          |     | frame. This value is                            |
|                          |          |     | frame. This value is calculated using the SPICE |

|            |         | the SPICE kernels.                             |
|------------|---------|--|
|            |         | Calculations will be based on                  |
|            |         | the MIDOBS time. SPICE                         |
|            |         | quaternion standard is that                    |
|            |         | Q0 is the scalar value.                        |
|            |         |  |
|            |         | Spacecraft quaternion in                       |
|            |         | J2000 (q2= $\sin(\text{theta/2})$ )            |
|            |         | obtained from the NAIF                         |
|            |         | provided C kernel. SPICE                       |
|            |         | convention conversion to 3x3                   |
|            |         | matrix transforms vector in                    |
|            |         | spacecraft frame to J2000                      |
|            |         | frame. This value is                           |
|            |         | calculated using the SPICE                     |
|            |         |  |
|            |         | interface and numbers will be                  |
|            |         | accurate to the accuracy of                    |
|            |         | the SPICE kernels.                             |
|            |         | Calculations will be based on                  |
| geom.qsin2 | SC QY   | the MIDOBS time.                               |
|            | ->      | Spacecraft quaternion in                       |
|            |         | J2000 (q3= sin(theta/2))                       |
|            |         | obtained from the NAIF                         |
|            |         |  |
|            |         | provided C kernel. SPICE                       |
|            |         | convention conversion to 3x3                   |
|            |         | matrix transforms vector in                    |
|            |         | spacecraft frame to J2000                      |
|            |         | frame. This value is                           |
|            |         | calculated using the SPICE                     |
|            |         | interface and numbers will be                  |
|            |         | accurate to the accuracy of                    |
|            |         |  |
|            |         | the SPICE kernels.                             |
|            |         | Calculations will be based on                  |
| geom.qsin3 | SC_QZ   | the MIDOBS time.                               |
|            |         | Instrument quaternion in                       |
|            |         | $J2000 (q0 = \cos(t/2))$                       |
|            |         | obtained from the NAIF                         |
|            |         | provided C kernel. SPICE                       |
|            |         | convention conversion to 3x3                   |
|            |         |  |
|            |         | matrix transforms vector in                    |
|            |         | instrument frame to J2000                      |
|            |         | frame. This value is                           |
|            |         | calculated using the SPICE                     |
|            |         | interface and numbers will be                  |
|            |         | accurate to the accuracy of                    |
|            |         | the SPICE kernels.                             |
|            |         | Calculations will be based on                  |
|            |         |  |
|            |         | the MIDOBS time. SPICE                         |
|            |         | quaternion standard is that                    |
| geom.qcos0 | INST_QA | Q0 is the scalar value.                        |
|            |         | Instrument quaternion in                       |
|            |         | J2000 (q1= $\sin(\frac{1}{\sin(\frac{1}{2})})$ |
|            |         | obtained from the NAIF                         |
|            |         | provided C kernel. SPICE                       |
|            |         | convention conversion to 3x3                   |
|            |         |  |
|            |         | matrix transforms vector in                    |
|            |         | instrument frame to J2000                      |
|            |         | frame. This value is                           |
|            |         | calculated using the SPICE                     |
|            | 1       |  |
| geom.qsin1 | INST QX | interface and numbers will be                  |

| I                                | I        | 1   | accurate to the accuracy of                |
|----------------------------------|----------|-----|--|
|                                  |          |     | the SPICE kernels.                         |
|                                  |          |     | Calculations will be based on              |
|                                  |          |     | the MIDOBS time.                           |
|                                  |          |     | Instrument quaternion in                   |
|                                  |          |     | $J2000 (q2 = \sin(\frac{1}{2}))$           |
|                                  |          |     | obtained from the NAIF                     |
|                                  |          |     | provided C kernel. SPICE                   |
|                                  |          |     | convention conversion to 3x3               |
|                                  |          |     | matrix transforms vector in                |
|                                  |          |     | instrument frame to J2000                  |
|                                  |          |     | frame. This value is                       |
|                                  |          |     | calculated using the SPICE                 |
|                                  |          |     | interface and numbers will be              |
|                                  |          |     | accurate to the accuracy of                |
|                                  |          |     | the SPICE kernels.                         |
|                                  |          |     | Calculations will be based on              |
| geom.qsin2                       | INST QY  |     | the MIDOBS time.                           |
| 800111101112                     | 11,01_21 |     | Instrument quaternion in                   |
|                                  |          |     | J2000 (q3= sin(theta/2))                   |
|                                  |          |     | obtained from the NAIF                     |
|                                  |          |     | provided C kernel. SPICE                   |
|                                  |          |     | convention conversion to 3x3               |
|                                  |          |     | matrix transforms vector in                |
|                                  |          |     | instrument frame to J2000                  |
|                                  |          |     | frame. This value is                       |
|                                  |          |     | calculated using the SPICE                 |
|                                  |          |     | interface and numbers will be              |
|                                  |          |     | accurate to the accuracy of                |
|                                  |          |     | the SPICE kernels.                         |
|                                  |          |     | Calculations will be based on              |
| geom.qsin3                       | INST QZ  |     | the MIDOBS time.                           |
| geom.qsm3                        | 1NS1_QZ  |     | Coordinate type for reference              |
|                                  |          |     | pixel, values are either "RA               |
|                                  |          |     | -TAN" = gnomic or tangent                  |
|                                  |          |     | plane or "SIP" - simple                    |
| mission area stand               | CTYPE1   |     |  |
| mission_area.ctype1              | CITEI    |     | image polynomial.  Coordinate type for the |
|                                  |          |     |  |
|                                  |          |     | reference pixel, values are                |
|                                  |          |     | either "DECTAN" =                          |
|                                  |          |     | gnomic or tangent plane or                 |
| ii                               | CTVPE    |     | "SIP" - simple image                       |
| mission_area.ctype2              | CTYPE2   |     | polynomial.                                |
|                                  |          |     | Right ascension of the                     |
|                                  | CDIVITA  |     | reference pixel or boresight               |
| geom.right_ascension_angle       | CRVAL1   | Deg | vector in degrees.                         |
|                                  |          |     | Declination of reference                   |
|                                  | CDIVITA  |     | pixel or boresight vector in               |
| geom.declination_angle           | CRVAL2   | Deg | degrees.                                   |
|                                  |          |     | Units for the reference pixel              |
| mission_area.cunit1              | CUNIT1   | Deg | 1  |
|                                  |          |     | Units for the reference pixel              |
| mission_area.cunit2              | CUNIT2   | Deg | 2  |
|                                  |          |     | X coordinate pixel number of               |
|                                  |          |     | the boresight of the image of              |
|                                  |          |     | the reference point to which               |
|                                  |          |     | the projection and the                     |
| geom.horizontal.coordinate_pixel | CRPIX1   |     | rotation refer.                            |

|                                  | 1      | 1        | Y coordinate pixel number of                          |
|----------------------------------|--------|----------|---|
|                                  |        |          | the boresight of the image of                         |
|                                  |        |          | the reference point to which                          |
|                                  |        |          | the projection and the                                |
| eom.vertical.coordinate.pixel    | CRPIX2 |          | rotation refer.                                       |
| 1                                |        |          | Change in RA per pixel                                |
|                                  |        |          | along first axis (sample)                             |
| mission area.cd1 1               | CD1 1  | Deg      | evaluated at reference pixel                          |
|                                  |        |          | Change in RA per pixel                                |
|                                  |        |          | along second axis (line)                              |
| mission area.cd1 2               | CD1 2  | Deg      | evaluated at reference pixel                          |
|                                  |        |          | Change in DEC per pixel                               |
|                                  |        |          | along first axis (sample)                             |
| mission_area.cd2_1               | CD2_1  | Deg      | evaluated at reference pixel                          |
|                                  |        |          | Change in DEC per pixel                               |
|                                  |        |          | along second axis (line)                              |
| mission_area.cd2_2               | CD2_2  | Deg      | evaluated at reference pixel                          |
|                                  |        |          | Azimuth of the North polar                            |
|                                  |        |          | Axis of the target named in                           |
|                                  |        |          | FITS keyword BENNURDT                                 |
|                                  |        |          | (typically Bennu), positive                           |
|                                  |        |          | from the +NAXIS2 direction                            |
| mission area.bennana             | n/a    | Deg      | toward the +NAXIS1                                    |
| mission_area.oeimana             | 11/ a  | Deg      | direction; see also                                   |
|                                  |        |          | BENNURDQ for a statement                              |
|                                  |        |          | of the quality of this value;                         |
|                                  |        |          | assumes undistorted optics;                           |
|                                  |        |          | will be -999 if the calculation                       |
|                                  |        |          | fails   |
|                                  |        |          | Right Ascension of the                                |
|                                  |        |          | vector, expressed in the Earth                        |
|                                  |        |          | Mean Equator of the J2000                             |
|                                  |        |          | Epoch, from the ORX                                   |
| mission area.bennu ra            | n/a    | Deg      | spacecraft toward the target                          |
|                                  |        |          | named in FITS keyword                                 |
|                                  |        |          | BENNURDT (typically                                   |
|                                  |        |          | Bennu); see also                                      |
|                                  |        |          | BENNURDQ for a statement of the quality of this value |
|                                  |        | +        | Declination of the vector,                            |
|                                  |        | 1        | expressed in the Earth Mean                           |
|                                  |        | 1        | Equator of the J2000 Epoch,                           |
|                                  |        | 1        | from the ORX spacecraft                               |
| mission area.bennu dec           | n/a    | Deg      | toward the target named in                            |
|                                  |        | 150      | FITS keyword BENNURDT                                 |
|                                  |        |          | (typically Bennu); see also                           |
|                                  |        |          | BENNURDQ for a statement                              |
|                                  |        |          | of the quality of this value                          |
|                                  |        | 1        | Approximate offset from                               |
|                                  |        | 1        | CRPIX1 pixel in +NAXIS1                               |
|                                  |        | 1        | direction of the location of                          |
|                                  |        |          | the center of the target                              |
| mission area honny mayis1 -fft   | 70/0   | Pixel    | named in FITS keyword                                 |
| mission_area.bennu_naxis1_offset | n/a    | rixei    | BENNURDT (typically                                   |
|                                  |        | 1        | Bennu); see also                                      |
|                                  |        | 1        | BENNURDQ for a statement                              |
|                                  |        | 1        | of the quality of this value;                         |
|                                  |        | <u> </u> | assumes undistorted optics                            |
|                                  |        |          | •   |

| mission_area.bennu_naxis2_offset | n/a      | Pixel | Approximate offset from CRPIX2 pixel in +NAXIS2 direction of the location of the center of the target named in FITS keyword BENNURDT (typically Bennu); see also BENNURDQ for a statement of the quality of this value; assumes undistorted optics   |
|----------------------------------|----------|-------|--|
| mission_area.bennu_radec_target  | n/a      |       | Target for the BENNURA, BENNUDEC, BENNUNX1 and BENNUNX2 FTIS keywords; typically, BENNU; may be NONE if the calculation failed. Target is not required to be in the field of view.   |
| mission_area.bennu_radec_quality | n/a      |       | (Quality: provenance) for the BENNURA, BENNUDEC, BENNUNX1, BENNUNX2 FITS keywords. This will be one of three values: (BEST: SPK), meaning the geometry was obtained from SPICE SP-Kernels; (POOR: osculating elements; +/-1E6km), meaning the geometry was obtained from osculating orbital elements of Bennu w.r.t the Sun, and will have uncertainties of order 1E6km; (NONE: FAILURE), meaning both the SPK and elements methods failed; the parentheses, (), are only delimiters here and not part of the quality:provenance values. |
| mission_area.radesys             | RADESYS  |       | International Celestial<br>Reference System (ICRS)   |
| geom.name                        | EQUINOX  |       | Epoch of mean equator and equinox (J2000)  |
| geom.celestial_north_clock_angle | ORIENTAT | Deg   | The angle (in degrees) between the image positive y-axis and celestial north.  |
| geom.spice_kernel_file_name      | META_KER |       | Metakernel that holds all the spice kernels used for processing  Quality of C-Kernel   |
| mission area.ckqual              | CKQUAL   |       | (nominal =  'RECONSTRUCT', contingency = 'PREDICT')  |
| mission_area.misspxls            | MISSPXLS |       | Count of pixels where data is missing in the image   |
| mission_area.checksum_result     | CHCKSUM  |       | The pass/fail state of the image check sum   |

| Spacecraft quatermion in 12000 (cg) cost/22) obtained at the ATT TIME, Quatermion values are telemetry values based on the ATT TIME parameter and may differ from the SPICE calculated quatermions at mid-observation time by several seconds. This value is to be used with caution as the telemetry values based on the ATT QUATI    Material of the corresponding image of the corresponding image. SPICE quatermion in 12000 (cg) is the sear value which is followed here. Spacecraft quatermion in 12000 (cg) is singularity and the SPICE calculated quatermion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quatermions in the several seconds. This value is to be used with caution as the timing many not be for the boresight of the corresponding image. Spacecraft quatermion in 12000 (cg) and the parameter and may differ from the SPICE calculated quatermion in 12000 (cg) and the ATT_TIME. Quatermion values are telemetry values based on the ATT_TIME quatermion in 12000 (cg) and the ATT_TIME quatermion in 12000 (cg) and the ATT_TIME quatermion in 12000 (cg) and the corresponding image. Spacecraft quatermion in 12000 (cg) and the corresponding image. Spacecraft quatermion in 12000 (cg) and the corresponding image. Spacecraft quatermion in 12000 (cg) and the corresponding image. Spacecraft quatermion in 12000 (cg) and the corresponding image. Spacecraft quatermion in 12000 (cg) and the corresponding image. Spacecraft quatermion in 12000 (cg) and the corresponding image. Spacecraft quatermion in 12000 (cg) and the corresponding image. Spacecraft quatermion in 12000 (cg) and the corresponding image. Spacecraft quatermion in 12000 (cg) and the corresponding image. Spacecraft quatermion in 12000 (cg) and the corresponding image. Spacecraft quatermion in 12000 (cg) and the corresponding image. Spacecraft quatermion in 12000 (cg) and the corresponding image. Spacecraft quatermion in 12000 (cg) and the corresponding image. Spacecraft quatermion in 12000 (cg) and the corresponding image.   |                           |              |                                |
|--|---------------------------|--------------|--------------------------------|
| JiDOO (q0 = cost(22) obtained at the ATT_TIME Quatermion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quatermions at mid-observation time by several seconds. This value is to be used with caution as the timing many not be for the borselight of the corresponding image. SPICE quatermion values are telemetry values based on the ATT_TIME. Quatermion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quatermion in JiDOO (q2 = sin(theat2)) obtained at the ATT_TIME parameter and may differ from the SPICE calculated quatermion in JiDOO (q2 = sin(theat2)) obtained at the ATT_TIME parameter and may differ from the SPICE calculated quatermion in JiDOO (q2 = sin(theat2)) obtained at the ATT_TIME parameter and may differ from the SPICE calculated quatermion in JiDOO (q2 = sin(theat2)) obtained at the ATT_TIME parameter and may differ from the SPICE calculated quatermions at mid-observation time by several seconds. This value is to be used with caution as the timing many not be for the borselight of the corresponding image.    Spaceraft quatermion of the parameter and may differ from the SPICE calculated quatermions at mid-observation time by several seconds. This value is to be used with caution as the timing many not be for the borselight of the corresponding image.    Spaceraft quatermion at mid-observation time by several seconds. This value is to be used with caution as the timing many not be for the borselight of the corresponding image.  |                           |              | Spacecraft quaternion in       |
| obtained at the ATT_TIME, Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the timing many not be for the boresight of the corresponding image. SPICE quaternion standard is that Q0 is the scalar value which is followed here:  ### ATTQUATO  ATTQUA |                           |              |                                |
| Quatermion values are telemetry values based on the ATT TIME parameter and may differ from the SPICE calculated quatermions at mid-observation time by several seconds. This value is to be used the cartion as the timing many not be for the boresight of the corresponding image. SPICE quatermion standard is that Q0 is the several value which is followed here.  ATTQUATO  ATTQUATO  ATTQUATO  ATTQUATI  ATTQUATION  ATTQUATI  ATTQUATION  ATTQUATI  AT |                           |              |                                |
| telemetry values based on the ATT TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the timing many not be for the boresight of the corresponding image. SPICE quaternion standard is that Q0 is the separation of the difference of the separation of the  |                           |              |                                |
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| mission_area.quaternion1  ATTQUAT1  Spacecraft quaternion in J2000 (q2= sin(theta/2))obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the timing many not be for the boresight of the corresponding image.  Spacecraft quaternion in J2000 (q3= sin(theta/2)) obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the   |                           |              |                                |
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| J2000 (q2= sin(theta/2))obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid- observation time by several seconds. This value is to be used with caution as the timing many not be for the boresight of the corresponding image.  Spacecraft quaternion in J2000 (q3= sin(theta/2)) obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the   | mission_area.quatermon1   | ATTQUATT     |                                |
| sin(theta/2))obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the timing many not be for the boresight of the corresponding image.  ATTQUAT2  ATTQUAT2  ATTQUAT2  ATTQUAT2  Spacecraft quaternion in J2000 (q3= sin(theta/2)) obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the   |                           |              |                                |
| ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at midobservation time by several seconds. This value is to be used with caution as the timing many not be for the boresight of the corresponding image.  ATTQUAT2  ATTQUATCA  ATTQUATC |                           |              |                                |
| values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid- observation time by several seconds. This value is to be used with caution as the timing many not be for the boresight of the corresponding image.  ATTQUAT2  ATTQUAT2  ATTQUAT2  Spacecraft quaternion in J2000 (q3= sin(theta/2)) obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the  |                           |              | sin(theta/2))obtained at the   |
| values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid- observation time by several seconds. This value is to be used with caution as the timing many not be for the boresight of the corresponding image.  ATTQUAT2  ATTQUAT2  ATTQUAT2  Spacecraft quaternion in J2000 (q3= sin(theta/2)) obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the  |                           |              | ATT TIME. Quaternion           |
| based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid- observation time by several seconds. This value is to be used with caution as the timing many not be for the boresight of the corresponding image.  ATTQUAT2  ATTQUAT2  ATTQUAT2  Spacecraft quaternion in J2000 (q3= sin(theta/2)) obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the  |                           |              |                                |
| parameter and may differ from the SPICE calculated quaternions at midobservation time by several seconds. This value is to be used with caution as the timing many not be for the boresight of the corresponding image.  ATTQUAT2  ATTQUAT2  ATTQUAT2  Spacecraft quaternion in J2000 (q3= sin(theta/2)) obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternion time by several seconds. This value is to be used with caution as the  |                           |              |                                |
| from the SPICE calculated quaternions at midobservation time by several seconds. This value is to be used with caution as the timing many not be for the boresight of the corresponding image.  ATTQUAT2  ATTQUAT2  Spacecraft quaternion in J2000 (q3= sin(theta/2)) obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the   |                           |              |                                |
| quaternions at midobservation time by several seconds. This value is to be used with caution as the timing many not be for the boresight of the corresponding image.  ATTQUAT2  ATTQUAT2  Spacecraft quaternion in J2000 (q3= sin(theta/2)) obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the   |                           |              |                                |
| observation time by several seconds. This value is to be used with caution as the timing many not be for the boresight of the corresponding image.  ATTQUAT2  ATTQUAT2  Spacecraft quaternion in J2000 (q3= sin(theta/2)) obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the   |                           |              |                                |
| seconds. This value is to be used with caution as the timing many not be for the boresight of the corresponding image.  ATTQUAT2  Spacecraft quaternion in J2000 (q3= sin(theta/2)) obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the   |                           |              |                                |
| seconds. This value is to be used with caution as the timing many not be for the boresight of the corresponding image.  ATTQUAT2  Spacecraft quaternion in J2000 (q3= sin(theta/2)) obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the   |                           |              | observation time by several    |
| used with caution as the timing many not be for the boresight of the corresponding image.  ATTQUAT2  Spacecraft quaternion in J2000 (q3= sin(theta/2)) obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the  |                           |              |                                |
| timing many not be for the boresight of the corresponding image.  Spacecraft quaternion in J2000 (q3= sin(theta/2)) obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the   |                           |              |                                |
| boresight of the corresponding image.  Spacecraft quaternion in J2000 (q3= sin(theta/2)) obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the  |                           |              |                                |
| mission_area.quaternion2  ATTQUAT2  Spacecraft quaternion in J2000 (q3= sin(theta/2)) obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the   |                           |              |                                |
| Spacecraft quaternion in J2000 (q3= sin(theta/2)) obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the   |                           |              |                                |
| J2000 (q3= sin(theta/2)) obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the  | mission_area.quaternion2  | ATTQUAT2     |                                |
| J2000 (q3= sin(theta/2)) obtained at the ATT_TIME. Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the  |                           |              | Spacecraft quaternion in       |
| obtained at the ATT_TIME.  Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the  |                           |              |                                |
| Quaternion values are telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the   |                           |              |                                |
| telemetry values based on the ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the   |                           |              |                                |
| ATT_TIME parameter and may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the   |                           |              |                                |
| may differ from the SPICE calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the  |                           |              |                                |
| calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the  |                           |              |                                |
| calculated quaternions at mid-observation time by several seconds. This value is to be used with caution as the  |                           |              | may differ from the SPICE      |
| mid-observation time by several seconds. This value is to be used with caution as the  |                           |              |                                |
| several seconds. This value is to be used with caution as the  |                           |              |                                |
| to be used with caution as the   |                           |              |                                |
|  |                           |              |                                |
| mission_area.quaternion3   ATTQUAT3   timing many not be for the   |                           | A TOTAL A TO |                                |
|  | mission_area.quaternion3  | ATTQUAT3     | timing many not be for the     |

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| 1                             | 1               | I        | boresight of the                                       |
|-------------------------------|-----------------|----------|--|
|                               |                 |          | corresponding image.                                   |
|                               |                 |          |  |
|                               |                 |          | The x-axis component of the                            |
|                               |                 |          | spacecraft angular                                     |
|                               |                 |          | momentum rate in spacecraft                            |
|                               |                 |          | body frame at the                                      |
| mission area.att rate x       | ATTRATE1        |          | ATŤ TIME   |
|                               |                 |          | The y-axis component of the                            |
|                               |                 |          | spacecraft angular                                     |
|                               |                 |          | momentum rate in spacecraft                            |
|                               | 4 TOTAL 4 TOTAL |          | body frame at the                                      |
| mission_area.att_rate_y       | ATTRATE2        |          | ATT_TIME   |
|                               |                 |          | The z-axis component of the                            |
|                               |                 |          | spacecraft angular<br>momentum rate in spacecraft      |
|                               |                 |          | body frame at the                                      |
| mission area att rate z       | ATTRATE3        |          | ATT TIME   |
|                               |                 |          | The spacecraft clock time                              |
|                               |                 |          | when attitude data was                                 |
| mission_area.att_time         | ATT_TIME        |          | collected  |
|                               |                 |          | Touch and Go Camera Suite                              |
|                               |                 |          | Powered On Digital Video                               |
|                               |                 |          | Recorder: 0 =  |
|                               | DI ID ON        |          | TAGCAMS_DVR_1,1 =                                      |
| mission_area.powered_on_dvr   | DVRON           |          | TAGCAMS_DVR_2  |
| mission_area.image_len        | IMG_LEN         | Bytes    | Recorded image length                                  |
|                               |                 |          | TAGCAMS commanded                                      |
| mission_area.sequence_id      | TCSEQID         |          | sequence identifier                                    |
|                               |                 |          | TAGCAMS Commanded                                      |
| mission_area.image_id         | TCIMGID         |          | Image Identifier                                       |
|                               |                 |          | TAGCAMS Mini Header                                    |
|                               |                 |          | spacecraft clock time in                               |
|                               |                 |          | seconds. The spacecraft clock time tag is the start of |
|                               |                 |          | exposure for line 1 of the                             |
|                               |                 |          | image plus or minus 0.1                                |
| mission area.seconds          | TCSCLKS         | Sec      | seconds.   |
|                               |                 |          | TAGCAMS Mini Header                                    |
|                               |                 |          | spacecraft clock time in sub-                          |
|                               |                 |          | seconds. The spacecraft                                |
|                               |                 |          | clock time tag is the start of                         |
|                               |                 |          | exposure for line 1 of the                             |
| mining and alternate          | TOGOT MOS       | 3 4:11.  | image plus or minus 0.1                                |
| mission_area.subseconds       | TCSCLKSS        | Millisec | seconds.  TAGCAMS Command 8-bit                        |
| mission area.img cmd opcode   | TCOPCODE        |          | Operations Code (Opcode)                               |
| inission_area.inig_ema_opeoue | TCOTCODE        |          | Specifies which camera to                              |
|                               |                 |          | use. Legal values are 1 (for                           |
|                               |                 |          | both DVR-4s) and 2 (for the                            |
|                               |                 |          | DVR-4 with StowCam                                     |
| mission_area.img_cmd_cam      | TCCAM           |          | attached).   |
|                               |                 |          | The sequence identifier to                             |
|                               | _               |          | assign to these images. Legal                          |
| mission_area.img_cmd_seq      | TCSEQ           |          | values are 1 to 255.                                   |

|                               |        |       | The number of images to                               |
|-------------------------------|--------|-------|---|
|                               |        |       | acquire in this sequence.                             |
| mission_area.img_cmd_num_imgs | TCN_IM |       | Legal values are 1 to 255.                            |
|                               |        |       | The commanded 16-bit                                  |
|                               |        |       | exposure time for each                                |
|                               |        |       | image. For values from 0 to                           |
|                               |        |       | 5000, exposure time is in                             |
|                               |        |       | units of 0.1 msec, providing                          |
|                               |        |       | exposures of 0 to 0.5                                 |
|                               |        |       | seconds. From 5001 to 65535, exposure time is in      |
|                               |        |       | units of 0.5msec with an                              |
|                               |        |       | offset of 0.5 seconds, so that                        |
|                               |        |       | the maximum exposure time                             |
|                               |        |       | is 0.5e-3*(65535-5000)                                |
| mission_area.img_cmd_exp      | TCEXP  |       | +0.5 = 30.7675 seconds.                               |
|                               | Tesm   |       | The 16-bit time interval                              |
|                               |        |       | between images in seconds,                            |
|                               |        |       | from 0 (as fast as possible) to                       |
| mission area.img cmd int      | TCINT  | Sec   | 65535 seconds.  |
|                               |        |       | The starting X of the area of                         |
|                               |        |       | the image sensor to read, in                          |
|                               |        |       | multiples of 16 pixels. If 0 is                       |
|                               |        |       | used for all values, then the                         |
|                               |        |       | full frame is read. (Full                             |
|                               |        |       | frames = $2592x1944$ pixels                           |
|                               |        |       | with dark pixels off,                                 |
|                               |        |       | 2752x2004 pixels with dark                            |
|                               |        |       | pixels on. Other values are                           |
|                               |        |       | valid if windowing is                                 |
|                               |        |       | enabled) Reading outside the                          |
|                               | maarr  | p: 1  | bounds of the image array is                          |
| mission_area.img_cmd_sx       | TCSX   | Pixel | illegal.  |
|                               |        |       | The starting Y of the area of                         |
|                               |        |       | the image sensor to read, in                          |
|                               |        |       | multiples of 16 pixels. If 0 is                       |
|                               |        |       | used for all values, then the                         |
|                               |        |       | full frame is read (Full frame                        |
|                               |        |       | = 2592x1944 pixels with<br>dark pixels off, 2752x2004 |
|                               |        |       | pixels with dark pixels on.                           |
|                               |        |       | Other values are valid if                             |
|                               |        |       | windowing is enabled)                                 |
|                               |        |       | Reading outside the bounds                            |
| mission area.img cmd sy       | TCSY   | Pixel | of the image array is illegal.                        |
|                               |        |       | The starting Z of the area of                         |
|                               |        |       | the image sensor to read, in                          |
|                               |        |       | multiples of 16 pixels. If 0 is                       |
|                               |        |       | used for all values, then the                         |
|                               |        |       | full frame is read (Full frame                        |
|                               |        |       | $= 2592 \times 1944$ pixels with                      |
|                               |        |       | dark pixels off, 2752x2004                            |
|                               |        |       | pixels with dark pixels on.                           |
|                               |        |       | Other values are valid if                             |
|                               |        |       | windowing is enabled)                                 |
|                               |        |       | Reading outside the bounds                            |
| mission_area.img_cmd_w        | TCW    | Pixel | of the image array is illegal.                        |

| mission area.img cmd h         | ТСН     | The starting height of the area of the image sensor to read, in multiples of 16 pixels. If 0 is used for all values, then the full frame is read (Full frame = 2592x1944 pixels with dark pixels off, 2752x2004 pixels with dark pixels on. Other values are valid if windowing is enabled) Reading outside the bounds of the image Pixel                              |
|--------------------------------|---------|--|
| mission_area.mig_cmd_n         | ICH     |  |
| mission area.img cmd mode      | TCMODE  | 0- 12-to-8 bit companding on/off; 1-3 companding mode (0 = square root, 1-7 linear divide by exp2(N-1)); 4- dark pixels on/off; test pattern control (on=1/off=0); 6 enable additional sensor register settings (enable=1/disable=0); 7 reserved   |
| mission_area.mig_emd_mode      | TEMODE  | The sensor analog gain   |
| mission area ima and gain      | TOGAIN  | value. Legal values are 8-31 and 40-63. Values from 8 to 31 specify gains in the range 1 to 3.875 in steps of 0.125; values from 40 to 63 specify gain in the range 2 to 7.75 in steps of 0.25. Gains in the range (2, 4) are commanded  |
| mission_area.img_cmd_gain      | TCGAIN  | Factor using the first range   |
| mission area.img cmd subsample | TCSSMPL | The subsampling to use when acquiring the image. Legal values for this field are 0 (no subsampling), 1 (bin 2x2), 4 (bin 4x4), and 16-23 (skip (N-14)*(N-14)).   |
|                                |         | Take image hardware image compression mode. 0 means no compression; 0xff means lossless compression; 1-99 means JPEG compression quality N, 4:2:2 color subsampling if applicable; 101-199 means JPEG compression quality N, 4:4:4 color subsampling if applicable. Hardware compression is only available in 8-bit mode; if companding is turned off, this value must |
| mission_area.img_cmd_compress  | TCCOMP  | be 0.  |
|                                |         | Records the pixel depth of the transmitted image. If the image was initially acquired as 12 bits, then it can be sent  |
| mission_area.bpp               | TCBPP   | bits/pixel as 12 bits or 8 bits. If the  |
|                                | •       |  |

|                                    |          |       | image was initially acquired as 8 bits, then this value is ignored, and the output is always 8 bits. The companding mode is only |
|------------------------------------|----------|-------|--|
|                                    |          |       | significant if companding is   |
|                                    |          |       | turned on. Keyword may be blank if the transmit-image-   |
|                                    |          |       | raw command was used to  |
|                                    |          |       | acquire image.  Compression mode. 0 means  |
|                                    |          |       | no compression; 0xff means   |
|                                    |          |       | lossless compression; 1-99   |
|                                    |          |       | means JPEG compression<br>quality N, grayscale; 101-199  |
|                                    |          |       | means JPEG compression   |
|                                    |          |       | quality N, 4:2:2 color   |
|                                    |          |       | subsampling. JPEG  |
|                                    |          |       | compression can only be applied to 8-bit data. Note  |
|                                    |          |       | that if a color camera's image   |
|                                    |          |       | is compressed as grayscale,  |
|                                    |          |       | compression efficiency may   |
|                                    |          |       | be degraded. Keyword may be blank if the transmit-   |
|                                    |          |       | image-raw command was  |
| mission area.trans cmd compression | TCTRCOMP |       | used to acquire image.   |
|                                    |          |       | Turns 2x2 summing on or  |
|                                    |          |       | off. Legal values are 1 (no  |
|                                    |          |       | summing) and 2 (2x2  |
|                                    |          |       | summing.) Summing can only be applied to 8-bit data.   |
|                                    |          |       | Images with both   |
|                                    |          |       | compression and summing  |
|                                    |          |       | selected will be compressed  |
|                                    |          |       | but not summed. Keyword  |
|                                    |          |       | may be blank if the transmit-<br>image-raw command was   |
| mission_area.trans_cmd_summing     | TCSUM    |       | used to acquire image.   |
|                                    |          |       | Camera head temperature in   |
| mission_area.camera_head_temp      | TCCHTEMP | DN    | digital number (if available).   |
| mission_area.spare                 | TCSPR    |       | Spare  |
|                                    | TODICZ   | D (   | initial size estimate for image  |
| mission_area.image_size_estimate   | TCINSZ   | Bytes | in flash   |

## 5.2.2 LO Status Data Product

The L0 status data product is the raw engineering data generated by TAGCAMS.

**Table 8 - L0 Status Data Product Fields** 

| Fields Name                      | Field  | Fields<br>Locations | Data Type    | Field<br>Length | Field<br>Length | Units        | Description   |
|----------------------------------|--------|---------------------|--------------|-----------------|-----------------|--------------|---|
|                                  | Number | (Start Byte)        |              | (bits)          | (bytes)         |              |   |
|                                  |        | (3.11.1)            |              |                 |                 |              | seconds portion of the timestamp  |
| seconds_raw                      | 1      | 1                   | UnsignedMSB4 | 32              | 4               | seconds      | of the status reading milliseconds portion of the   |
| subseconds_raw                   | 2      | 5                   | UnsignedByte | 8               | 1               | milliseconds | timestamp of the status reading   |
| spare0                           | 3      | 6                   | UnsignedByte | 8               | 1               | n/a          | spare   |
| spare1                           | 4      | 7                   | UnsignedByte | 8               | 1               | n/a          | spare   |
| command opcode                   | 5      | 8                   | UnsignedByte | 8               | 1               | n/a          | opcode of command producing<br>this packet, or 0x20 for status  |
| last opcode                      | 6      | 9                   | UnsignedMSB4 | 32              | 4               | n/a          | last opcode received  |
| valid_cmds_cnt                   | 7      | 13                  | UnsignedMSB4 | 32              | 4               | n/a          | valid commands received since<br>power-on   |
| rejected_cmds_cnt                | 8      | 17                  | UnsignedMSB4 | 32              | 4               | n/a          | rejected commands received since power-on   |
| inst_sw_ver                      | 9      | 21                  | UnsignedMSB4 | 32              | 4               | n/a          | instrument software version   |
| checksum                         | 10     | 25                  | UnsignedMSB4 | 32              | 4               | n/a          | checksum of instrument software   |
| num_bad_flash_blocks             | 11     | 29                  | UnsignedMSB4 | 32              | 4               | n/a          | number of bad flash blocks  |
| num_free_flash_blocks            | 12     | 33                  | UnsignedMSB4 | 32              | 4               | n/a          | number of free flash blocks   |
| total usable flash blocks        | 13     | 37                  | UnsignedMSB4 | 32              | 4               | n/a          | total number of usable flash<br>blocks  |
| num products in use              | 14     | 41                  | UnsignedMSB4 | 32              | 4               | n/a          | number of products in use   |
| num pages used                   | 15     | 45                  | UnsignedMSB4 | 32              | 4               | n/a          | number of flash pages used in products  |
| flash err ent                    | 16     | 49                  | UnsignedMSB4 | 32              | 4               | n/a          | flash error count in last built-in self-test  |
| dram errors                      | 17     | 53                  | UnsignedMSB4 | 32              | 4               | n/a          | Dynamic random access memory errors in last built-in self-test  |
| num_correctable_ecc_error        |        |                     |              |                 |                 |              | number of correctable error   |
| S                                | 18     | 57                  | UnsignedMSB4 | 32              | 4               | n/a          | correcting code (ECC) errors<br>number of sequences stored in   |
| num_sequences_stored             | 19     | 61                  | UnsignedMSB4 | 32              | 4               | n/a          | flash   |
| camera_statu                     | 20     | 65                  | UnsignedMSB4 | 32              | 4               | n/a          | camera status as bitmask  |
| num_uncorrectable_ecc_er<br>rors | 21     | 69                  | UnsignedMSB4 | 32              | 4               | n/a          | number of uncorrectable error correcting code (ECC) errors  |
| num camera head upsets           | 22     | 73                  | UnsignedMSB4 | 32              | 4               | n/a          | number of camera head command errors  |
| itum_eamera_neaa_apsets          | 22     | 7.5                 | Charghedwigh | 32              | •               | 11/ 4        | extended error code for last  |
| ext_err_code                     | 23     | 77                  | UnsignedMSB4 | 32              | 4               | n/a          | command ending in error   |
| text_data_checksum               | 24     | 81                  | UnsignedMSB4 | 32              | 4               | n/a          | text+data checksum  |
| flash_device_status              | 25     | 85                  | UnsignedMSB4 | 32              | 4               | n/a          | flash device status as bitmask  |
| dram_device_status               | 26     | 89                  | UnsignedMSB4 | 32              | 4               | n/a          | bitmask of dynamic random<br>access memory (DRAM) device<br>status from last built-in self-test,<br>nominal 0 |
| num flash op timouts             | 27     | 93                  | UnsignedMSB4 | 32              | 4               | n/a          | number of flash operation timeouts  |
| num_flash_op_timouts             | 28     | 97                  | UnsignedMSB4 | 32              | 4               | n/a          | execution time of last command in ticks, if measured  |
| bram ecc status register         | 29     | 101                 | UnsignedMSB4 | 32              | 4               | n/a          | Block random access memory<br>error correction code (BRAM<br>ECC) status register                             |

|                               |        | Fields       |                 | Field  | Field   |       |  |
|-------------------------------|--------|--------------|-----------------|--------|---------|-------|--|
| Fields Name                   | Field  | Locations    | Data Type       |        | Length  | Units | Description  |
|                               | Number | (Start Byte) | J F -           | (bits) | (bytes) |       |  |
| num spacewire timeouts<       | 30     | 105          | UnsignedMSB4    | 32     | 4       | n/a   | number of Spacewire timeouts   |
| num received char overru      | 50     | 103          | Olisighedivisb+ | 32     | _       | 11/α  | number of spacewire timeouts   |
| ns                            | 31     | 109          | UnsignedMSB4    | 32     | 4       | n/a   | overruns   |
| orrectable_bram_ecc_ers_c     | 32     | 113          | UnsignedMSB4    | 32     | 4       | n/a   | count of correctable Block<br>random access memory error<br>correction code (BRAM ECC)<br>errors |
|                               |        | 117          |                 | 32     |         |       |  |
| last_image_id                 | 33     | 11/          | UnsignedMSB4    | 32     | 4       | n/a   | last image identifier acquired   |
| num_images_acquired           | 34     | 121          | UnsignedMSB4    | 32     | 4       | n/a   | number of images acquired  |
| last_time_update_msg_tim<br>e | 35     | 125          | UnsignedMSB4    | 32     | 4       | n/a   | time of last time update message (seconds)   |
| num time ticks seen           | 36     | 129          | UnsignedMSB4    | 32     | 4       | n/a   | number time ticks seen   |
| fpga_logic_version            | 37     | 133          | UnsignedMSB4    | 32     | 4       | n/a   | FPGA (field-programmable gate array) logic version   |
| camera_0_current              | 38     | 137          | UnsignedMSB4    | 32     | 4       | DN    | camera 0 current   |
| camera_1_current              | 39     | 141          | UnsignedMSB4    | 32     | 4       | DN    | camera 1 current   |
| camera_2_current              | 40     | 145          | UnsignedMSB4    | 32     | 4       | DN    | camera 2 current   |
| camera_3_current              | 41     | 149          | UnsignedMSB4    | 32     | 4       | DN    | camera 3 current   |
| camera_0_voltage              | 42     | 153          | UnsignedMSB4    | 32     | 4       | DN    | camera 0 voltage   |
| camera_1_voltage              | 43     | 157          | UnsignedMSB4    | 32     | 4       | DN    | camera 1 voltage   |
| camera_2_voltage              | 44     | 161          | UnsignedMSB4    | 32     | 4       | DN    | camera 2 voltage   |
| camera_3_voltage              | 45     | 165          | UnsignedMSB4    | 32     | 4       | DN    | camera 3 voltage   |
| camera_0_temp                 | 46     | 169          | UnsignedMSB4    | 32     | 4       | DN    | camera 0 temperature   |
| camera_1_temp                 | 47     | 173          | UnsignedMSB4    | 32     | 4       | DN    | camera 1 temperature   |
| camera_2_temp                 | 48     | 177          | UnsignedMSB4    | 32     | 4       | DN    | camera 2 temperature   |
| camera_3_temp                 | 49     | 181          | UnsignedMSB4    | 32     | 4       | DN    | camera 3 temperature   |
| dvr_pos1_2v                   | 50     | 185          | UnsignedMSB4    | 32     | 4       | DN    | Digital Video Recorder plus 1.2<br>Volt or internal temperature<br>monito.r                      |
| dvr_pos2_5v                   | 51     | 189          | UnsignedMSB4    | 32     | 4       | DN    | Digital video recorder plus 2.5 volt monito.r  |
| dvr_pos3_3v                   | 52     | 193          | UnsignedMSB4    | 32     | 4       | DN    | Digital video recorder plus 3.3 volt monitor.  |
| dvr_pos5v                     | 53     | 197          | UnsignedMSB4    | 32     | 4       | DN    | Digital video recorder plus 5 volt monitor.  |

## **5.2.3 L1 Status Data Product**

The L1 Status data product contains engineering values converted from DNs to physical units.

| Fields Name                      | Field  | Field<br>Location | Data Type    | Field<br>Length | Field<br>Length | Units        | Description   |
|----------------------------------|--------|-------------------|--------------|-----------------|-----------------|--------------|---|
|                                  | Number | (Start<br>Byte)   |              | (bits)          | (bytes)         |              |   |
| seconds_raw                      | 1      | 1                 | UnsignedMSB4 | 32              | 4               | seconds      | seconds portion of the timestamp of the status reading            |
| subseconds_raw                   | 2      | 5                 | UnsignedByte | 8               | 1               | milliseconds | milliseconds portion of the<br>timestamp of the status<br>reading |
| spare0                           | 3      | 6                 | UnsignedByte | 8               | 1               | n/a          | spare   |
| spare1                           | 4      | 7                 | UnsignedByte | 8               | 1               | n/a          | spare   |
| command opcode                   | 5      | 8                 | UnsignedByte | 8               | 1               | n/a          | opcode of command<br>producing this packet, or<br>0x20 for status |
| last_opcode                      | 6      | 9                 | UnsignedMSB4 | 32              | 4               | n/a          | last opcode received  |
| valid_cmds_cnt                   | 7      | 13                | UnsignedMSB4 | 32              | 4               | n/a          | valid commands received since power-on                            |
| rejected cmds cnt                | 8      | 17                | UnsignedMSB4 | 32              | 4               | n/a          | rejected commands<br>received since power-on                      |
| inst sw ver                      | 9      | 21                | UnsignedMSB4 | 32              | 4               | n/a          | instrument software version                                       |
| checksum                         | 10     | 25                | UnsignedMSB4 | 32              | 4               | n/a          | checksum of instrument software                                   |
| num_bad_flash_blo<br>cks         | 11     | 29                | UnsignedMSB4 | 32              | 4               | n/a          | number of bad flash blocks  |
| num_free_flash_blo<br>cks        | 12     | 33                | UnsignedMSB4 | 32              | 4               | n/a          | number of free flash blocks                                       |
| total_usable_flash_b<br>locks    | 13     | 37                | UnsignedMSB4 | 32              | 4               | n/a          | total number of usable flash blocks                               |
| num_products_in_u<br>se          | 14     | 41                | UnsignedMSB4 | 32              | 4               | n/a          | number of products in use   |
| num_pages_used                   | 15     | 45                | UnsignedMSB4 | 32              | 4               | n/a          | number of flash pages used in products                            |
| flash_err_cnt                    | 16     | 49                | UnsignedMSB4 | 32              | 4               | n/a          | flash error count in last<br>BIST                                 |
| dram_errors                      | 17     | 53                | UnsignedMSB4 | 32              | 4               | n/a          | DRAM errors in last BIST  |
| num_correctable_ec<br>c_errors   | 18     | 57                | UnsignedMSB4 | 32              | 4               | n/a          | number of correctable ECC errors                                  |
| num_sequences_sto<br>red         | 19     | 61                | UnsignedMSB4 | 32              | 4               | n/a          | number of sequences stored in flash                               |
| camera_statu                     | 20     | 65                | UnsignedMSB4 | 32              | 4               | n/a          | camera status as bitmask  |
| num_uncorrectable_<br>ecc_errors | 21     | 69                | UnsignedMSB4 | 32              | 4               | n/a          | number of uncorrectable ECC errors                                |
| num_camera_head_<br>upsets       | 22     | 73                | UnsignedMSB4 | 32              | 4               | n/a          | number of camera head command errors                              |
| ext_err_code                     | 23     | 77                | UnsignedMSB4 | 32              | 4               | n/a          | extended error code for last command ending in error              |
| text_data_checksum               | 24     | 81                | UnsignedMSB4 | 32              | 4               | n/a          | text+data checksum  |
| flash_device_status              | 25     | 85                | UnsignedMSB4 | 32              | 4               | n/a          | flash device status as<br>bitmask                                 |

#### Field Field Field Fields Name Field Units Data Type Description Location Length Length (Start Number (bits) (bytes) Byte) bitmask of DRAM device 89 UnsignedMSB4 32 26 4 n/a status from last BIST, dram device status nominal 0 num flash op timo number of flash operation 27 93 UnsignedMSB4 32 4 n/a timeouts execution time of last num\_flash\_op\_timo 28 97 UnsignedMSB4 32 4 n/a command in ticks, if measured bram\_ecc\_status\_re 29 101 UnsignedMSB4 32 4 BRAM ECC status register n/a gister num\_spacewire\_tim number of Spacewire 30 105 32 4 UnsignedMSB4 n/a eouts< timeouts num received char number of receiver 32 31 109 UnsignedMSB4 4 n/a overruns character overruns orrectable bram ec count of correctable 32 113 UnsignedMSB4 32 4 n/a c ers ent BRAM ECC errors last image identifier 33 117 UnsignedMSB4 32 4 n/a last image id acquired num\_images\_acquir UnsignedMSB4 32 34 121 4 n/a number of images acquired time of last time update last\_time\_update\_m 32 4 35 125 UnsignedMSB4 n/a message (seconds) sg\_time num\_time\_ticks\_see 36 129 32 4 number of lost time ticks UnsignedMSB4 n/a37 133 32 UnsignedMSB4 4 FPGA logic version n/a fpga\_logic\_version IEEE754MSBSingl 137 32 4 38 Milliamps camera 0 current camera 0 current IEEE754MSBSingl 39 141 32 4 Milliamps camera 1 current camera 1 current IEEE754MSBSingl 40 145 32 4 Milliamps camera 2 current camera 2 current IEEE754MSBSingl 41 149 32 4 Milliamps camera 3 current camera\_3\_current IEEE754MSBSingl 42 153 32 4 Volts camera 0 voltage camera 0 voltage IEEE754MSBSingl 157 32 4 Volts 43 camera 1 voltage camera\_1\_voltage IEEE754MSBSingl 44 161 32 4 Volts camera 2 voltage camera\_2\_voltage IEEE754MSBSingl 32 4 45 165 Volts camera 3 voltage camera\_3\_voltage IEEE754MSBSingl 46 169 32 4 degC camera 0 temperature camera 0 temp IEEE754MSBSingl 47 173 32 4 degC camera 1 temperature camera\_1\_temp IEEE754MSBSingl 48 177 32 4 degC camera 2 temperature camera 2 temp IEEE754MSBSingl 181 32 49 4 degC camera 3 temperature camera\_3\_temp Digital Video Recorder IEEE754MSBSingl 50 185 32 4 Volts plus 1.2 Volt or internal dvr pos1 2v temperature monitor IEEE754MSBSingl Digital video recorder plus 189 51 32 4 Volts dvr\_pos2\_5v 2.5 volt monitor IEEE754MSBSingl Digital video recorder plus 52 193 32 4 Volts dvr pos3 3v 3.3 volt monitor IEEE754MSBSingl Digital video recorder plus 53 197 32 4 Volts dvr\_pos5v 5 volt monitor

For the current, voltage, and temperature channels, the DN-to-EU mapping will be of the form eu = a\*dn+b, where a and b will be supplied for each channel after characterization.

For the voltage channels, the nominal conversion is a=610.352e-6, b=0, EU is in volts.

For the current channels, the nominal conversion is a=0.1525879, b=0, EU is in milliamps.

For the temperature channels, the nominal conversion is a=0.15259, b=-273.14, EU in degrees C. Characterization of the flight units shows that the best-fit value of b is -275.02 for NavCam (primary DVR) and-273.43 for NFTCam and StowCam (secondary DVR.)

## 5.3 Label and Header Description

All data products are produced with PDS4 compliant detached XML labels. Examples of these labels can be found in Appendix 7.5. FITS headers are described in Section 5.2

## 6 Applicable Software

The following sections describe display software that may be used to examine, display, or analyze the NavCam data products.

## 6.1 Utility Programs

At the current time, the OSIRIS-REx project has no plans to release any mission specific utility programs. As most TAGCAMS data products are FITS formatted files, any viewer with the capability of reading FITS files can be used to view the data products. Some examples of these viewers are IDL, J-Mars (<a href="http://jmars.asu.edu/download">http://jmars.asu.edu/download</a>), and FV. A complete list of FITS viewers can be found at <a href="http://fits.gsfc.nasa.gov/fits\_viewer.html">http://fits.gsfc.nasa.gov/fits\_viewer.html</a>.

## 6.2 Applicable PDS Software Tools

The PDS supplies a number of software tools that can be used in conjunction with PDS data products. Please refer to the PDS4 software website (<a href="http://pds.nasa.gov/pds4/software/index.shtml">http://pds.nasa.gov/pds4/software/index.shtml</a>) for additional information on these tools.

## 6.3 Software Distribution and Update Procedure

As the OSIRIS-REx project will not be providing software, this section is not applicable.

## 7 Appendices

## 7.1 Definitions of Data Processing Levels

Table 10 shows the comparison of OSIRIS-REx, NASA and CODMAC data processing levels. The OSIRIS-REx team generally uses descriptions when classifying data rather than data levels.

**Table 9 - Definitions of Data Processing Levels** 

| OSIRIS-<br>REx           | NASA           | CODMAC             | Description   |
|--------------------------|----------------|--------------------|---|
|                          | Packet<br>data | Raw - Level 1      | Telemetry data stream as received at the ground station, with science and engineering data embedded.  |
| Level 0 -<br>Raw         | Level 0        | Edited - Level 2   | Instrument science data (e.g., raw voltages, counts) at full resolution, time ordered, with duplicates and transmission errors removed.   |
| Level 1-<br>Uncalibrated | Level<br>1A    | Calibrated - Level | NASA Level 0 data that have been located in space and may have been transformed (e.g., calibrated, rearranged) in a reversible manner and packaged with needed ancillary and auxiliary data (e.g., radiances with the calibration equations applied). |
| Level 2 -<br>Calibrated  | Level<br>1B    | Resampled - Level  | Irreversibly transformed (e.g., resampled, remapped, calibrated) values of the instrument measurements (e.g., radiances, magnetic field strength).  |
| Level 3 -<br>Processed   | Level<br>1C    | Derived - Level 5  | NASA Level 1A or 1B data that have been resampled and mapped onto uniform space-time grids. The data are calibrated (i.e., radiometrically corrected) and may have additional corrections applied (e.g., terrain correction).                         |
| Level 4 -<br>Derived     | Level 2        | Derived - Level 5  | Geophysical parameters, generally derived from Level 1 data, and located in space and time commensurate with instrument location, pointing, and sampling.   |

| Level 4 -<br>Derived | Level 3 | Derived - Level 5 | Geophysical parameters mapped onto uniform space-time grids. |
|----------------------|---------|-------------------|--|
|----------------------|---------|-------------------|--|

| OSIRIS-REx Data I | Product Level Definitions   |
|-------------------|---|
| Level             | Definition  |
| OREx Level 0      | Telemetry. Raw instrument data reconstructed from telemetry with header and ancillary information appended. Appended header and |
|                   | ancillary data is data necessary for further processing.  |
| OREx Level 1      | Uncalibrated. Data in one of the fundamental structures.  |
| OREx Level 2      | Reversibly calibrated. Data in units proportional to physical units.  |
|                   | Since PDS allows offsets and scaling factors in its array and table   |
|                   | structures, this would be the minimum level capable of satisfying the   |
|                   | "in physical units" requirement.  |
| OREx Level 3      | Irreversibly processed. Higher-level products from a single source  |
|                   | that cannot be losslessly converted back to the lower-level products  |
|                   | from which they were derived. These might also satisfy the "in  |
|                   | physical units" requirement.  |
| OREx Level 4      | Derived data. Products created by combining data from more than   |
|                   | one source (instrument, observer, etc.).  |

## 7.2 Example PDS Labels

Example labels can be found in the TAGCAMS bundle document collection in a sub-directory named "example\_labels". There are example labels for each type of TAGCAMS data product.